such a lightweight structural component. As more superplastic alloys such as 7475 aluminum become commercially available, this type of investigation will undoubtedly attract more attention from the researchers in the area of lightweight structures.

References

¹Hamilton, C. H., "Superplastic Forming Makes Molded Parts from Sheet Metal Alloys," Industrial Research and Development, Vol. 25, Dec. 1983, pp. 72-76.

²Ueng, C. E. S. and Liu, T. L., "Least Weight Design of Superplastically Formed Sandwich Cores," Proceedings of the 2nd ASCE Engineering Mechanics Specialty Conference, North Carolina State University, Raleigh, May 1977, pp. 440-443.

³Ueng, C. E. S. and Liu, T. L., "Least Weight Analysis of a New Sandwich Core," *Proceedings of 14th Annual Meeting of the* Society of Engineering Science, Lehigh University, Bethlehem, Pa., Nov. 1977, pp. 273-283.

⁴Ueng, C. E. S. and Liu, T. L., "Optimization of New Light-Weight Sandwich Core," Proceedings of International Conference on Light-Weight Shell and Spatial Structures for Normal and Seismic Zones (sponsored by International Association of Shell and Spatial Structures), Alma-Ata, USSR, Sept. 1977, pp. 375-383.

⁵Fiacco, A. V. and McCormick, G. P., Nonlinear Programming: Sequential Unconstrained Minimization Techniques, John Wiley & Sons, New York, 1968.

⁶Ueng, C. E. S. and Kim, T. D., "Shear Modulus of Core Materials with Arbitrary Polygonal Shape," Computers and Struc-

tures, Vol. 16, No. 1, 1983, pp. 21-25.

7"Mechanical Properties of Hexcel Honeycomb Materials," Hexcel Co., Dublin, Calif., Rept. TSB 120, 1974.

Radiative Effect on the Conjugated **Forced Convection-Conduction** Analysis of Heat Transfer in a Plate Fin

Fue-Sang Lien* and Cha'o-Kuang Chen† National Cheng Kung University Tainan, Taiwan, China and

John W. Cleaver‡

University of Liverpool, Liverpool, England, U.K.

Nomenclature

= specific heat at constant pressure

= temperature difference parameter, $T_{\infty}/(T_0-T_{\infty})$

= reduced stream function

= dimensional and dimensionless modified heattransfer coefficients

= fluid and fin thermal conductivity

= fin length

= convection-conduction parameter, $(kL/k_f\delta)Re_L^{1/2}$

= conduction-radiation parameter, $k\beta^*/4\sigma(T_0-T_\infty)^3$

= Prandtl number, ν/α

= radiative heat flux, $(-4\sigma/3\beta^*)(\partial T^4/\partial y)$, for opti-

cally thick limit approximation

 $\displaystyle \underset{T,\,T_f}{Q}$ = overall heat-transfer rate

= fluid and fin temperature

†Professor, Department of Mechanical Engineering.

‡Senior Lecturer, Department of Mechanical Engineering.

 T_0 = root temperature

= velocity components in x and y directions, respecu, v

tively

= coordinates system shown in Fig. 1 x, y

= thermal diffusivity α

β* = extinction coefficient = half-thickness of the fin

δ

 ξ,η = pseudosimilarity variables

= dimensionless fluid and fin temperatures θ, θ_f

= kinematic viscosity

= density of fluid

= Stefan-Boltzmann constant

= stream function

Subscripts

= condition at wall

 ∞ = condition at freestream

Introduction

I N the conventional heat transfer analysis of fins, it is standard practice to assume that the heat transfer coefficient for convection at the fin surfaces is uniform. However, there is evidence in the literature demonstrating that the heat-transfer coefficient can experience substantial variations along the fin surfaces. 1,2

This Note is concerned with fins that transfer heat to a surrounding fluid by forced convection and radiation. The heat-transfer coefficient along the fin is not prescribed, but rather solved in advance from the boundary-layer convection flow. Therefore, the modified local heat transfer coefficient is determined by a highly coupled interaction among the fin conduction, radiation, and fluid convection flow.

The problem to be analyzed here is illustrated in Fig. 1. A vertical fin of length L and thickness 2δ is extended from a wall at temperature T_0 and situated in a uniform freestream having temperature T_{∞} and velocity u_{∞} . The optically thicken limit approximation for the radiative flux is assumed, and the tip for the fin is insulated.

Governing Equations

Consider now a vertical fin parallel to a uniform freestream. Let x and y denote, respectively, the streamwise and normal coordinates. The conservation equations for a laminar boundary layer over a vertical fin is

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \tag{1}$$

$$u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = v\frac{\partial^2 u}{\partial y^2}$$
 (2)

$$u\frac{\partial T}{\partial x} + v\frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2} - \frac{1}{\rho C_p} \frac{\partial q^r}{\partial y}$$
 (3)

subjected to the following boundary conditions:

$$u=v=0$$
, $T=T_w$, at $y=0$

$$u = u_{\infty}$$
, $T = T_{\infty}$, as $y \to \infty$ (4)

Introducing pseudosimilarity variables (ξ, η) with a reduced stream function $f(\eta)$ and a dimensionless temperature

$$\xi = \frac{x}{L}, \quad \eta = \left(\frac{y}{L}\right) Re_L^{1/2}/\xi^{1/2}, \qquad f(\eta) = \psi \frac{(x,y)}{(u_\infty L \xi \nu)^{1/2}},$$

$$\theta = \frac{T - T_{\infty}}{T_0 - T_{\infty}} \tag{5}$$

Received Jan. 23, 1984; revision received Sept. 24, 1984. Copyright © American Institute of Aeronautics and Astronautics, Inc., 1985. All rights reserved.

^{*}Graduate Student, Department of Mechanical Engineering.

where the stream function ψ satisfies the continuity equation with $u = \partial \psi / \partial y$ and $v = -\partial \psi / \partial x$.

When the transformation is applied to Eqs. (2-4) and the optically thick limit approximation for q^r is assumed, the governing system becomes

$$f''' + \frac{1}{2}ff'' = 0 \tag{6}$$

$$Pr^{-1}\theta'' + \frac{1}{2}f\theta' + \frac{4[(\theta + C_T)^3\theta']'}{3PrN} = \xi f'\frac{\partial\theta}{\partial\xi}$$
 (7)

$$f = f' = 0$$
, $\theta_w = [T_w(x) - T_\infty]/(T_0 - T_\infty)$, at $\eta = 0$
 $f' = I$, $\theta = 0$, as $\eta \to \infty$ (8)

where the primes denote partial differentiation with respect to n.

Considering a very thin fin, it is reasonable to assume a one-dimensional model for the fin temperature along the x-coordinate. The fin conservation energy equation is

$$\frac{\mathrm{d}^2 T_f}{\mathrm{d}x^2} = \frac{h^*(x)}{k_f \delta} (T_f - T_\infty) \tag{9}$$

The associated boundary conditions are

$$T_f = T_0$$
, at $x = L$ and $\frac{\mathrm{d}T_f}{\mathrm{d}x} = 0$, at $x = 0$ (10)

The basic coupling between the fin and the convective flow is expressed by the requirement that the fin and fluid temperatures and heat fluxes be continuous at the fin/fluid interface at all x positions.

$$T_f = T_w$$
 and $-k \frac{\partial T}{\partial y} + q^r = h^*(T_f - T_\infty)$, at $y = 0$ (11)

Equation (9) was recast in dimensionless form by the substitution

$$\xi = x/L, \ \theta_f = (T_f - T_{\infty})/(T_0 - T_{\infty})$$
 (12)

Equation (9) then becomes

$$\frac{\mathrm{d}^2 \theta_f}{\mathrm{d} \xi^2} = N_c \hat{h}^* \theta_f \tag{13}$$

where N_c is the convection-conduction parameter $[N_c = (kL)/k_f\delta)Re_L^{f/2}]$ and \hat{h}^* is a dimensionless local heat transfer coefficient with radiative effect for the current cycle of the iteration

$$\hat{h}^* = -\left(I + \frac{4(\theta + C_T)^3}{3N}\right) \frac{\partial \theta}{\partial n} / (\theta_f \xi^{1/2}), \text{ at } \eta = 0$$
 (14)

The method to be used here involves a succession of consecutive solutions for the fin and boundary layer. Initially, the boundary-layer equations are solved subject to an assumed fin temperature distribution that satisfies the thermal boundary conditions, and from this solution \hat{h}^* is evaluated. By substituting \hat{h}^* into the fin energy equation, we get a new value for θ_f . This new θ_f is imposed as the fin surface boundary condition for the governing equations. This procedure of alternately solving the boundary-layer problem and fin conduction problem was continued until convergence was attained.

Numerical Procedure

The solutions of the boundary-layer equation were obtained by an implicit finite-difference method devised by

Cebeci and Bradshaw.³ According to this method, Eqs. (6) and (7) are first written in the first-order equation by introducing new known functions. The functions and their derivatives in the first-order equations are then approximated by centered difference quotients and averaged at the midpoints of net rectangles to yield finite difference equations. The resulting nonlinear difference equations are finally solved using Newton's method. The fin conduction equation is also written in the finite difference form. For small ξ , a finer ξ subdivision was needed for the boundary-layer and heat conduction equations. The present method uses 45 points in the streamwise direction and 61 grid points in the cross-stream direction.

Results and Discussion

Numerical results of the overall rate of heat transfer Q from the fin can be obtained from the heat conducted from the wall into the fin base at $\xi = 1$ or by integrating the local heat flux at the fin surface. The corresponding heat flux values of these two methods are found to be in agreement. They may be expressed in dimensionless form as follows in Eq. (15)

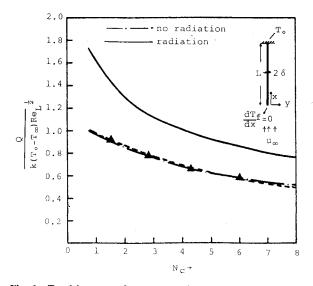


Fig. 1 Total heat-transfer rate. ---, simple model; A, Sparrow.4

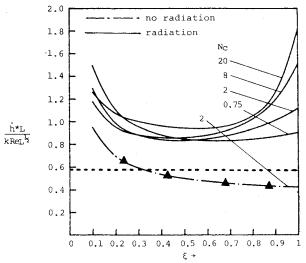


Fig. 2 The modified heat-transfer coefficients along the plate fin for N=1, $C_T=0.5$, Pr=0.7, and various values of Nc. ---, simple model; \blacktriangle , Sparrow.⁴

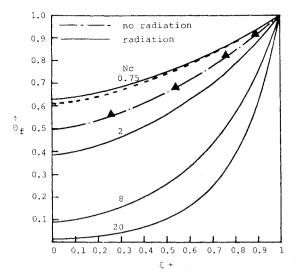


Fig. 3 The temperature distribution along the plate fin for N=1, $C_T = 0.5$, and various values of Nc. ---, simple model; \triangle , Sparrow.⁴

$$\frac{Q}{k(T_0 - T_\infty)Re_L^{1/2}} = 2\int_0^1 \left(-\frac{\partial\theta}{\partial\eta}\right)$$

$$\times \left[I + \frac{4(\theta + C_T)^3}{3N}\right] / \xi^{1/2} d\xi, \text{ at } \eta = 0$$
(15)

or

$$\frac{Q}{k(T_0 - T_\infty)Re_L^{\frac{1}{2}}} = \frac{2}{N_c} \left. \frac{\mathrm{d}\theta_f}{\mathrm{d}\xi} \right|_{\xi = 1} \tag{16}$$

The results of the overall rate of heat transfer Q from the fin are shown in Fig. 1 for various values of N_c . Figure 1 indicates that the overall heat-transfer rate of the fin with radiative effect is higher than that of the fin without radiative effect. The agreement of the results from the simple model with those of Ref. 4, and the special case $(q^r = 0)$ in this paper is good.

The distributions of the modified local heat-transfer coefficient \hat{h}^* for forced convection and radiation along the fin with different N_c are shown in Fig. 2. The modified heat transfer coefficient can be obtained from Eq. (14). For higher values of N_c , the fin is more nonisothermal. Although the local heat-transfer coefficient without radiative effects⁴ monotonically decreases in the fluid flow direction, the modified local heat transfer coefficients computed in this paper with fixed radiative effect do not vary monotonically. In the direction from tip to base, those coefficients decrease at first, attain a minimum, and then increase. Figure 2 also shows that the local heat transfer coefficient with radiative effect is higher than that without radiative effect.

Figure 3 presents fin temperature distributions for forced convection flow with radiative effect. In this figure, it is shown that the fin temperature decreases monotonically from the root to tip. The figure also shows that the larger values of N_c give rise to larger fin temperature variations and the fin temperature without radiative effect is always higher than that with radiative effect.

Conclusion

The analysis in this Note has yielded the results of physical fin for forced convection flow with radiative effect under the optically thick limit approximation. The agreement of the results for special case q' = 0 with Ref. 4 is remarkable.

References

¹Sparrow, E. M., Baliga, B. R., and Patankar, S. V., "Forced Convection Heat Transfer from a Shrouded Fin Array with and without Tip Clearance," ASME Journal of Heat Transfer, Vol. 100, 1978, pp. 572-579.

²Stachiewicz, J. W., "Effect of Variation of Local Film Coefficients on Fin Performance," ASME Journal of Heat Transfer, Vol. 91, 1969, pp. 21-26.

Cebeci, T. and Bradshaw, P., Momentum Transfer in Boundary

Layers, Hemisphere Publishing, Washington, D.C., 1977.

⁴Sparrow, E. M. and Chyu, M. K., "Conjugated Forced Convection-Conduction Analysis of Heat Transfer in a Plate Fin," ASME Journal of Heat Transfer, Vol. 104, 1982, pp. 204-206.

Improved Free Convective Heat-Transfer Correlations in the Near-Critical Region

Saeed Ghorbani-Tari* and Afshin J. Ghajar† Oklahoma State University, Stillwater, Oklahoma

Nomenclature

= specific heat at constant pressure

= mean-integral heat capacity, $\bar{C}_p = (i_w - i_\infty)/(T_w - T_\infty)$

= diameter

= gravitational constant

= Grashof number, $(gD^3\rho^2/\mu^2)[(\rho_\infty - \rho_w)/\rho_w]$

= heat-transfer coefficient

= enthalpy

k = thermal conductivity

= characteristic length

Nu =Nusselt number, hL/k

 $Pr = Prandtl number, \mu C_p/k$

 $Ra = \text{Rayleigh number}, Gr \cdot Pr$

= temperature

= absolute viscosity

= density

Subscripts

= evaluated at the film temperature,

 $T_f\!=(T_w+T_\infty)/2$

= evaluated at the wall temperature

= evaluated at the freestream (bulk) temperature

Introduction

VARIETY of correlations have been developed for A prediction of free convective heat-transfer rates¹⁻⁷ (see Table 1). Discrepancies exist between these correlations when they are applied in the near-critical region. The most significant sources of discrepancy appear to be the influences of 1) the particular reference temperature used in the evaluation of the physical properties, 2) the particular physical properties selected in reducing the dimensional experimental data to the dimensionless variables, and 3) the differences in values of the physical properties used by the various investigators.

Presented as Paper 84-1705 at the AIAA 19th Thermophysics Conference, Snowmass, CO, June 25-28, 1984; received Oct. 4, 1984; revision received Feb. 8, 1985. Copyright © American Institute of Aeronautics and Astronautics, Inc., 1985. All rights reserved.

^{*}Graduate Student, School of Mechanical and Aerospace Engineering.

[†]Associate Professor, School of Mechanical and Aerospace Engineering. Member AIAA.